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MOTION OF THE TROPICAL STORM

YOSHITAKA NAMIKAWA

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MOTION OF THE TROPICAL STORM

* * * * * * *

Yoshitaka Namikawa



MOTION OF THE TROPICAL STORM

bу

Yoshitaka Namikawa
Lieutenant, Japanese Maritime
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Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN METEOROLOGY

United States Naval Postgraduate School Monterey, California

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Thesis

by

Yoshitaka Namikawa

This work is accepted as fulfilling the thesis requirements for the decree of MASTER OF SCIENCE

IN

METEOROLOGY

from the

United States Naval Postgraduate School



ABSTRACT

Equations for the motion of a tropical storm are derived by assuming that the wind field consists of the vortex motion superimposed on a basic current.

The differential equations are integrated numerically for a variety of typical pressure fields, namely, a constant pressure radient representing straight parallel isobars; a high pressure cell, represented by an elliptical paraboloid; and a "col", represented by a hyperbolic paraboloid.

The writer wishes to express his appreciation for the assistance and encouragement given him in this investigation by Professor G. J. Haltiner of the U. S. Naval Postgraduate School.



TABLE OF CONTENTS

Section	Title	Page
1	Introduction	1
2	Development of the Equation of Motion	3
3	Computations and Solutions	10
4	Verification	21
5	Conclusions	28
6	Bibliography	29



LIST OF ILLUSTRATIONS

Figure		Pare
1.	Illustration for Blaton's formula	4
2.	Illustration for the Coriolis parameter as a function of the latitude.	6
3.	Illustration for the tropical vortex	7
4.	Illustration for sign of the cardinal direction	9
5.	A representation of a high pressure cell by an elliptical paraboloid	11
6.	A representation of a "ccl" by a hyperbolic paraboloid	12
7.	Paths of a vortex for several orientations of straight parallel isobars	14
8.	Paths of a vortex for several initial velocities and basic pressure gradients in the case of straight parallel isobars	15
٥.	The paths of a vortex as a function of the initial latitude	16
10.	Paths of a vortex for varying vortex size and intensity	17
11.	The path of a vortex for an elliptic basic pressure pattern	18
12.	Paths of a vortex for a hyperbolic basic pressure pattern	20
13.	The first computation for typhoon "Clara" of Nov. 1950, I	22
14.	The second computation for typhcon "Clara" of Nov. 1950, II	24
15.	The third computation for typhoon "Clara" of Nov. 1950, III	25
16.	A computation for typhoon "Jean" of Oct. 1956	27



	Table						Pase
	I	The	data	for	Figure	7	13
	II	The	data	for	Figure	8	15
	III	The	data	for	Figure	9	16
	IV	The	data	for	Figure	10	17
	V	The	data	for	Figure	11	18
	VI	The	data	for	Fi ure	12	19
	VII	The	aata	for	Figure	13	21
7	/III	The	data	for	Figure	16	26



1. INTRODUCTION.

Many methods for the prognosis of the motion of tropical storms have been attempted. They may be divided into two main classes.

The first method is based on kinematical considerations and the use of the pressure tendency as developed by Elliott [2] and Petterssen [6]. Murakami, Masuda and Arakawa[5] adapted this technique to quantitative prediction.

The second method is based on dynamical considerations which consider the tropical storm as the vortex in the atmosphere. Yeh 9 analyzed the external and internal forces acting on Rankine model vortex, based on the Blasius law by using complex potential. As the solution of his equation, he found the circular path without the basic motion, and the trochoidal path with the basic motion which has uniform velocity.

Kuo 4, Takeuchi 8 analyzed the forces by using a stream function for the basic flow which has constant vorticity. Kasahara 3 developed the idea of stream function for numerical prediction based on the barotropic model. Syono 7 derived the equation of motion for the vortex in a non-uniform pressure field from some what more general considerations, and illustrated oscillatory motion of vortex.

The approach in this paper is similar to that of Syono but uses a more complicated model of the vortex. Also, for numerical predicting purposes the variability of the Coriolis factor with latitude is taken into account in the equations.

Because of the difficulties in obtaining analytical



National Cash Refister 102A electronic computer. A half-hour time increment is sufficient to integrate the equations for this purpose. About 40 minutes are needed to compute a 24 hour path. Moreover—three typical basic pressure fields are utilized in this paper, namely strai ht isobars, an elliptical high pressure cell, and a col. These pressure systems are represented mathematically by the following quadratic surfaces; a plane, an elliptic paraboloid and a hyperbolic paraboloid. The last one is an especially interesting case because it relates to the recurvature of the tropical storm.

The application of these solutions for numerical perdiction is not covered completely in this paper. However by taking the variability of basic pressure field, size and intensity of storm as a function of time and space, the method shows promise of fair accuracy.



2. Development of the equation of motion

Neglecting the vertical motion, friction force and earth's curvature, and taking a cartesian coordinate system with x and y positive eastward and northward, U and V are x and y components of velocity and the equations of motion for a air particle on the rotating earth are

$$\frac{dU}{dt} = -\frac{1}{P} \frac{\partial P}{\partial x} + \int V$$

$$\frac{dV}{dt} = -\frac{1}{P} \frac{\partial P}{\partial y} - \int U$$
(1)

where

 $f = 2\Omega \sin \phi$

 ϕ ; the latitude

 Ω ; the angular velocity of rotating earth

?; the density of air

P ; the atmospheric pressure

We shall separate this motion into the basic and rotating motion. Assuming air density is uniform everywhere,

let
$$P = p + p'$$

 $\Box = u + u'$
 $\nabla = v + v'$
(2)

where P, U, V represent the total motion; p, u, v, the basic; and p', u', v', the rotating motion.

Substituting (2) into (1) we obtain

$$\frac{d\nabla}{dt} = -\frac{1}{\rho} \frac{\partial P}{\partial x} - \frac{1}{\rho} \frac{\partial P}{\partial x} + fv + fv'$$

$$\frac{d\nabla}{dt} = -\frac{1}{\rho} \frac{\partial P}{\partial y} - \frac{1}{\rho} \frac{\partial P}{\partial y} - fu - fu'$$
(3)



We shall assume now that the rotating circular vortex moves without deformation and that the gradient wind approximation may be taken to represent the pressure contribution of the vortex as follows

$$\frac{1}{P} \frac{\partial P'}{\partial T} = \frac{\mathcal{V}_T^2}{T_t} + \int \mathcal{V}_T \tag{4}$$

where \mathbf{v}_{τ} is the tangential velocity and \mathbf{r}_{t} is radius of the particle trajectory.

Blaton's formula states

$$\Gamma_{t} = \Gamma_{s} \left(1 - \frac{C}{V_{T}} \cos \psi \right)^{-1} \tag{5}$$

Here r_s is radius of streamline, and c is velocity of system, and ψ is angle between c and ψ .

From Fig. 1

$$\psi = \left(\frac{\pi}{2} + \Theta\right) - \Theta_{\tau} = \frac{\pi}{2} - \left(\Theta_{\tau} - \Theta\right)$$

thus

$$\cos \psi = \sin \theta_T \cdot \cos \theta - \cos \theta_T \cdot \sin \theta \tag{6}$$

and

$$C \sin \Theta_{\tau} = C_{\gamma}$$

$$C \cos \Theta_{\tau} = C_{\gamma}$$
(7)

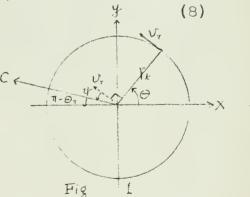
Combining (5), (6) and (7) leads to

$$T_t = T_s \left[1 - \frac{1}{v_t} \left(C_y \cos \Theta - C_x \sin \Theta \right) \right]^{-1}$$

No simple mathematical

function has been found to represent the wind distribution in the entire tropical storm.

Syono, Yeh, and others have





used a Rankine vortex with velocity distribution, $\frac{\mathcal{V}_{\tau}}{r}$ = constant, for the whole area of the storm. Takahashi introduced the empirical formulas

$$V_{\tau}$$
 = constant at the outside of storm V_{τ} | constant in the outer region (about within r = 500 kms) V_{τ} | constant in the inner region (about within r = 40 kms)

In this paper, the Rankine model will be taken for inner region and a hyperbolic wind distribution for outer region, following [1], namely:

$$v_r = \omega r$$
 ; $o \leqslant r \leqslant a$ (9)

$$V_{\tau} = k_{f}$$
 ; $\alpha \leqslant \Gamma \leqslant \Gamma$, (91)

ω is the angular velocity of rotating vortex, K is constant, a and r are the radii of the inner region and outer toundaries, respectively.

From (4)
$$\frac{1 \frac{\partial f'}{\partial x}}{= \cos \theta \cdot \frac{1 \frac{\partial f}{\partial x}}{= \sin \theta \cdot \frac{1}{\rho} \frac{\partial f}{\partial x}} = \cos \theta \cdot \frac{1}{\rho} \frac{\partial f}{\partial x} = \sin \theta \cdot \frac{1}{\rho} \frac{\partial f}{\partial x} = \cos \theta \cdot \frac{1}{\rho} \frac{\partial f}{\partial x} = \cos$$

$$-\frac{1}{\rho}\frac{\partial p'}{\partial X} = -\cos\theta \cdot \omega^2 f_s + \omega(C_y \cos^2\theta - C_x \sin\theta \cdot \cos\theta) - \cos\theta \omega f_s f$$

$$; c \leq r \leq \alpha$$
(10)

and =
$$-\cos\theta \cdot \frac{K^2}{l_s^2} + \frac{K}{l_s^2} (C_y \cos\theta - C_x \sin\theta \cdot \cos\theta) - \cos\theta \cdot \frac{Kf}{l_s^2}$$
 (101)

$$-\frac{1}{p}\frac{\partial p'}{\partial y} = -\sin\theta \cdot \omega^2 \Gamma_5 + \omega(C_y \cos\theta \cdot \sin\theta - C_x \sin^2\theta) - \sin\theta \omega \Gamma_5$$
 (11)

$$-\frac{1}{p}\frac{2f}{2y} = -\sin\theta \cdot w f_s + \omega(Cyas\theta \cdot \sin\theta - C_x \sin^2\theta) - \sin\theta w f_s f$$
and
$$= -\sin\theta \cdot \frac{K}{f_s^2} + \frac{K}{f_s^2}(C_y \cos\theta \cdot \sin\theta - C_x \sin^2\theta) - \sin\theta \cdot K f_s$$

$$= -\sin\theta \cdot \frac{K}{f_s^2} + \frac{K}{f_s^2}(C_y \cos\theta \cdot \sin\theta - C_x \sin^2\theta) - \sin\theta \cdot K f_s$$

$$= -\sin\theta \cdot \frac{K}{f_s^2} + \frac{K}{f_s^2}(C_y \cos\theta \cdot \sin\theta - C_x \sin^2\theta) - \sin\theta \cdot K f_s$$

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$$= -\sin\theta \cdot \frac{K}{f_s^2} + \frac{K}{f_s^2}(C_y \cos\theta \cdot \sin\theta - C_x \sin^2\theta) - \sin\theta \cdot K f_s$$

$$= -\sin\theta \cdot \frac{K}{f_s^2} + \frac{K}{f_s^2}(C_y \cos\theta \cdot \sin\theta - C_x \sin^2\theta) + \cos\theta \cdot \sin\theta - C_x \sin\theta \cdot K f_s$$

$$= -\sin\theta \cdot \frac{K}{f_s^2} + \frac{K}{f_s^2}(C_y \cos\theta \cdot \sin\theta - C_x \sin\theta - C$$



Later, r will be used instead of r_s .

Now $f=2\Omega\sin\varphi$ is a Coriolis parameter and is a function of the latitude. Let the latitude at center of vortex be φ , and R, r the radii of earth and vortex, respectively

The latitude at any point in the vortex ', r, as measured from the vortex center may be represented as follows:

$$\phi = \phi_6 + d\phi = \phi_6 + \frac{r \sin \theta}{R}$$
, $\sin \phi = \sin (\phi_6 + \frac{r \sin \theta}{R})$

as shown in Fig. 1 and Fig. 2.

Expanding $\cos \frac{r \sin \theta}{R}$ and $\sin \frac{r \sin \theta}{R}$ in series form and neglecting terms of higher order than the second gives

$$Sin\phi = Sin\phi_{s} \cdot cos(\frac{\Gamma sin\theta}{R}) + cos\phi_{g} \cdot sin(\frac{\Gamma sin\theta}{R})$$

$$= Sin\phi_{s}(1 - \frac{\Gamma^{2} sin^{2}\theta}{2R^{2}}) + cos\phi_{g} \cdot (\frac{\Gamma sin\theta}{R})$$
(12)

Next integrating equation (3) over the whole area of vortex, which is shown in Fig 3, we obtain

$$\frac{1}{S} \int \frac{dQ}{dt} dS = -\frac{1}{S} \int \frac{1}{P} \frac{\partial P}{\partial x} dS - \frac{1}{S} \int \frac{1}{P} \frac{\partial P}{\partial x} dS + \frac{1}{S} \int \frac{1}{P} \frac{\partial P$$

Now, let

$$-\frac{1}{p}\frac{\partial k}{\partial x} = b_{x}, \quad -\frac{1}{p}\frac{\partial k}{\partial y} = b_{x}$$

$$C_{x} = \frac{1}{s}\sqrt[4]{u}ds = \frac{1}{s}\sqrt[4]{u}ds$$

$$C_{x} = \frac{1}{s}\sqrt[4]{v}ds = \frac{1}{s}\sqrt[4]{v}ds$$

Integrating the first term

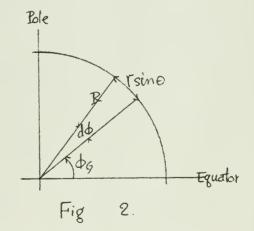
on the right side of equation

(13) yields

$$\mathcal{J} b_x ds = \pi t_1^2 \overline{b}_x$$

$$\mathcal{J} b_y ds = \pi t_1^2 \overline{b}_y$$

where \overline{b}_x , \overline{b}_y , are mean values.





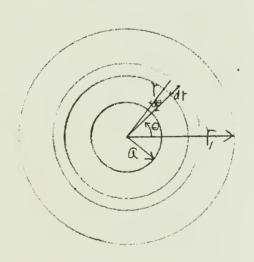
Furthermore, by use of equations (10) and (11)

$$-\sqrt[4]{\frac{3p'}{p \ni x}} ds = \int_{0}^{\pi} \int_{0}^{\pi} \left\{ -\cos \theta \cdot \omega^{2} \Gamma + \omega \left(C_{y} \cos \theta - C_{x} \sin \theta \cdot \cos \theta \right) \right\} ds$$

$$+ \int_{0}^{\pi} \int_{a}^{r} \left\{ -\cos \theta \frac{k^{2}}{t^{3}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta - C_{x} \sin \theta \cdot \cos \theta \right) \right\} ds - \sqrt[4]{\cos \theta} \int_{a}^{\pi} ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \left[C_{y} \cdot \pi \right] ds$$

$$= \frac{\omega}{2} a^{2} C_{y} \cdot \pi + K \ln \frac{r}{a} \cdot C_{y} \cdot \pi - \int_{a}^{\pi} \int_{a}^{r} \int_{a}^{r} \left[C_{y} \cos \theta \cdot \sin \theta - C_{x} \sin \theta \right] ds$$

$$+ \int_{0}^{\pi} \int_{a}^{r} \left[-\sin \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{x} \sin \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[-\sin \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{x} \sin \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[-\sin \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{x} \sin \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[-\sin \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{x} \sin^{2} \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{x} \sin^{2} \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{y} \sin^{2} \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{y} \sin^{2} \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{y} \sin^{2} \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{y} \sin^{2} \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \frac{K^{2}}{r^{2}} + \frac{K}{r^{2}} \left(C_{y} \cos \theta \cdot \sin \theta - C_{y} \sin^{2} \theta \right) \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \int_{a}^{r} \left[\cos \theta \cdot \cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \left[\cos \theta \cdot \cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \left[\cos \theta \cdot \cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \left[\cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \left[\cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \left[\cos^{2} \theta \right] ds - \sqrt[4]{\sin \theta} \int_{a}^{r} \left[\cos^{2$$



 $-9 \text{ fuds} = -C_{x} \cdot 2\Omega \sin \phi_{6} \cdot S$

Fig 3

Therefore the equations of motion become

$$\frac{dC_x}{dt} = 2\Omega \sin \phi_g \cdot C_x + \overline{b}_x + \frac{\alpha^2}{2\Gamma_i^2} \omega \cdot C_y + \frac{K}{\Gamma_i^2} \ln \frac{V_i}{\alpha} \cdot C_y$$

$$\frac{dC_x}{dt} = -2\Omega \sin \phi_g \cdot C_x + \overline{b}_y - \frac{\alpha^2}{2\Gamma_i^2} \omega \cdot C_x - \frac{K}{\Gamma_i^2} \ln \frac{\Gamma_i}{\alpha} \cdot C_x$$
(14)



Since
$$\phi_g$$
 is a function of y ,

 $Sin\phi_g = Sin\phi_o + y \frac{\cos\phi_o}{R} + \cdots$
 $2\Omega Sin\phi_o \cdot Cy = 2\Omega Sin\phi_o \cdot Cy + 2\Omega \cos\phi_o \cdot R Cy$
 $-2\Omega Sin\phi_o \cdot C_x = -2\Omega Sin\phi_o \cdot C_x - 2\Omega \cos\phi_o \cdot R C_x$

where ϕ_0 corresponds to y = 0. Channes in the annular velocity of the vortex may be approximated by assuming the conservation of absolute vorticity of the vortex after Syono [7].

$$\omega = \omega_0 + \Omega \left(\sin \phi_0 - \sin \phi \right)$$

$$\sin \phi \doteq \sin \phi_0 \left(1 - \frac{1}{2} \frac{y^2}{R^2} \right) + \cos \phi_0 \left(\frac{y}{R} \right)$$
(16)

Therefore
$$\frac{\alpha^2}{2r_i^2}\omega C_y = \frac{\omega_o}{2r_i^2}\alpha^2 C_y + \frac{\Omega\alpha^2}{4r_i^2R^2}C_y \cdot y + \frac{\Omega\alpha^2}{2r_i^2R}C_y \cdot \cos\phi_o \cdot y$$

$$-\frac{\alpha^2}{2r_i^2}\omega C_x = -\frac{\omega_o}{2r_i^2}\alpha^2 C_x - \frac{\Omega\alpha^2}{4r_i^2R^2}C_x \cdot y^2 \sin\phi_o + \frac{\Omega\alpha^2}{2r_i^2R}C_x \cdot \cos\phi_o \cdot y$$

Finally, the equations of motion are $\frac{dC_x}{dt} = \overline{b}_x + 2\Omega \sin \phi_0 \cdot C_y + \frac{2\Omega \cos \phi_0}{R} \cdot C_y + \frac{\omega_0}{2\Gamma_1^2} \alpha^2 C_y + \frac{\Omega \alpha^2}{4\Gamma_1^2 R^2} C_y y^2 \sin \phi_0$ - 202 Cy. cosd. y + K ln 11. Cy

 $\frac{dC_{x}}{dt} = \overline{b}_{x} - 2\Omega \sin \phi_{o} \cdot C_{x} - \frac{2\Omega \cos \phi_{o}}{2} \cdot y \cdot C_{x} - \frac{\omega_{o}}{2} \cdot \alpha^{2} \cdot C_{x} - \frac{\Omega \alpha^{2}}{4\pi^{2}R^{2}} C_{x} y^{2} \sin \phi_{o}$

(17)

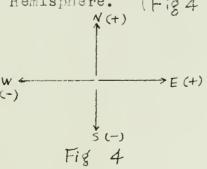
+ na Cx cospo. y - K ln 11 . Cx Next we shall check the magnitude of each term in the

right side of the equation (17). When x, y are taken as 1000 km, and $c_x = c_y = 5$ Kts, the magnitude of each term is as follows:

(1)
$$+\frac{\alpha^{2}}{2!} \omega_{0} C_{y}$$
, $-\frac{\alpha^{2}}{2!} \omega_{0} C_{x}$
(2) $+\frac{\Omega \alpha^{2}}{4!} \sin \phi_{0} C_{y} y^{2}$, $-\frac{\Omega \alpha^{2}}{4!} \sin \phi_{0} C_{x} y^{2}$
(3) $-\frac{\Omega \alpha^{2}}{2!} \cos \phi_{0} C_{y} y$, $+\frac{\Omega \alpha^{2}}{2!} \cos \phi_{0} C_{x} y$
(4) $+\frac{K}{1!} l_{1} \frac{Y_{1}}{\alpha} C_{y}$, $-\frac{K}{1!} l_{1} \frac{Y_{1}}{\alpha} C_{x}$
(5) $+2\Omega C_{y} \sin \phi_{0}$, $-2\Omega C_{x} \sin \phi_{0}$
(6) $+\frac{2\Omega \cos \phi_{0}}{R} \cdot y \cdot C_{y}$, $-\frac{2\Omega \cos \phi_{0}}{R} \cdot y \cdot C_{x}$
; $10^{2} \sim 10^{2} \text{ km hr}^{2}$



The terms \bar{b}_x , by are due to basic pressure field, and represent a contribution to the external steering. Terms (5) and (6) provide the Coriolis force for moving vortex. The other terms are due to internal character of the vortex itself. When c, has positive sin and the vortex has a northward component, terms (1), (2), (4) act toward east. but the term (3) acts westward. Te term (3) is relatively small compared with the terms (1), (4), hence the acceleration of the vortex is eastward. In the case of negative c,, the opposite effects take place. Similarly when c, has positive sign and vortex is moving eastward, the terms (1), (2), (4) act southward, while the small term (3) acts northward. The result is a net force southward. In the case of negative cx, the vortex tends to move northward. Comparison of terms (2) and (3), which are functions of y, shows that term (3) dominates term (2). Therefore, a northward movement with an increasing value of y will tend to make term (3) significant. When term (3) becomes sufficiently large, the northward movement tends to increase with positive cx . Also, the eastward movement decreases or westward movement increases with positive cy. All directions are considered in the Northern Hemisphere. (Fig.4)





3. Computation and Solutions

The numerical solution is obtained by the digital computer using the Runge-Kutta-Gill method which involves four iterations for each forward step. Half-hour time increments were found to be sufficient for computing the motion of tropical storms from this system of equations. About 40 minutes are needed for computing the 24 hour path on the CRC 102A digital computer at the U.S. Naval Postgraduate School.

Solutions are provided for a variety of conditions, which may be roughly divided into three cases, according to the following basic pressure patterns:

(A) A constant pressure gradient representing straight parallel isobars.

The terms \bar{b}_x , \bar{b}_y in the equation (17) are constant everywhere. Therefore \bar{b}_x , \bar{b}_y represent average values over the area of the vortex.

(B) A in h pressure cell, represented by an elliptical paraboloid.

A pressure surface in the vicinity of the subtropical high cell may be approximated by this shape, as shown in Fig.5, where the coordinates of the center of the high are \mathbf{x}_i , \mathbf{y}_i . Considering the conditions at some pressure level, such as the 700-mb or 500-mb surfaces, the height of the constant pressure surface \mathbf{z} is a function of \mathbf{x} and \mathbf{y} . The equation for the contour height may be written as follows,

$$(Z-Z_1) = A(X-X_1)^2 + B(y-y_1)^2$$
 (18)

where A and B are constants related to the slope of the

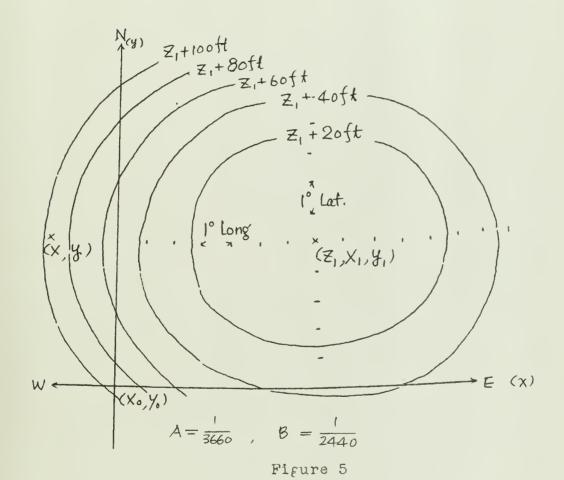


pressure surface. Here

$$b_{xg} = -9\frac{32}{5x} = -29A(x-x_1)$$

$$b_{yg} = -9\frac{32}{5} = -29B(y-y_1)$$
(19)

The symbol 6 is used for the b_x , b_y at the center of the vortex. If B is small but A is large, equation (18) gives an ellipsoid elongated along x axis. If A=B, there is a circular pressure pattern.



(C) A "col", represented by a hyperbolic paratoloid.

This pattern, which is closely associated with the recurvature of tropical storms, is shown in Figure 6.



The equation for this pattern may be written as follows:

$$(Z-Z_1) = A(X-X_1)^2 - B(Y-Y_1)^2$$
 (20)

where A, B are again constant, It follows that

$$b_{x6} = -9 \frac{2Z}{2X} = -29 A (X - X_1)$$

$$b_{y6} = -9 \frac{2Z}{2Y} = +29 B (Y - Y_1)$$
(21)

If B is small but A is large, there will be a strong trough.

On the contrary, a small A with large B indicates a weak

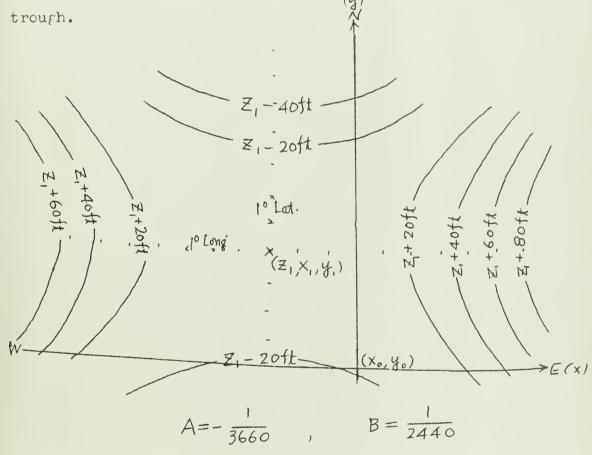


Figure 6

In case (2) and (3) b, and b, are not constart.



TAFLE I

$\omega = 2 \text{ hr}'$ $K = 1600 \text{ km}^2 \text{ hr}^{-1}$ $\phi_0 = 150 \text{N}$				r; = 300 km a = 40 km				
t		I	I		III		IV	
(hr)	C×	Cy	C _×	Cy	C _x	Cy	C _×	Cy
3	-15.9		-4.6	+17.2	+ 9.3	+15.9	+17.2	+4.1
6	- 6.4		+5.3	+12.7	+12.8	+ 6.1	+12.8	-5.1
9	+ 2.8	+ 8.0	+8.3	+ 2.3	+ 7.9	-3.1	+ 3.2	-8.2
12	+ 5.5	- 1.9	+2.9	-6.1	- 2.3	-5.5	-5.6	-3.2
15	-0.1	-11.1	-7.4	-8.1	-10.9	+2.3	-8.1	+6.6
18	-10.2	-12.4	-15.7	-1.8	-12.4	110.5	-2.6	+15.1
21	-18.3	-6.3	-16.6	+ 6.6	- 5.7	116.7	+7.5	+16.9
24	-19.4	+ 3.7	-9.4	+17.2	+ 5.1	+13.8	#15.8	110.4
27	-12.8	+11.5	+1.3	+16.0	+12.5	+10.7	H13.5	+0.0
30	- 2.6	+11.9	+8.3	+ 7.7	+11.1	+0.0	+9.1	-7.7
33	+ 4.5	+ 4.7	+6.5	- 3.0	+ 1.8	-6.3	-1.5	_7.2
36	+ 4.4	- 5.6	-2.8	-6.6	-8.8	-2.9	-8.1	+1.0
39	- 3.5	-12.3	-13.1	-5.3				
42	-13.7	-11.0	-18.0	+4.7				
45	-19.8	- 2.8	-12.5	+14.4				
48			-1.8	+17.1	entra analysis analysis			

Integrating b_x and b_y , given by equations (19) and (21), over the whole area of the vortex yields

Figure 7 and Table I show the path of vortex for several orientations of straight parallel isobars representing the basic pressure field. The models are taken with the data in Table 1. Curves I, II, III, IV are based on a basic pressure field corresponding to a 10 kt eostrophic wind, and a 10 kt initial velocity toward the directions of W, NW, N, NE respect-



ively. Each curve represents a cycloid with the same amplitude and period. When the path is sufficiently long, the curves may differ because of the contribution of term (3) in equation (17).

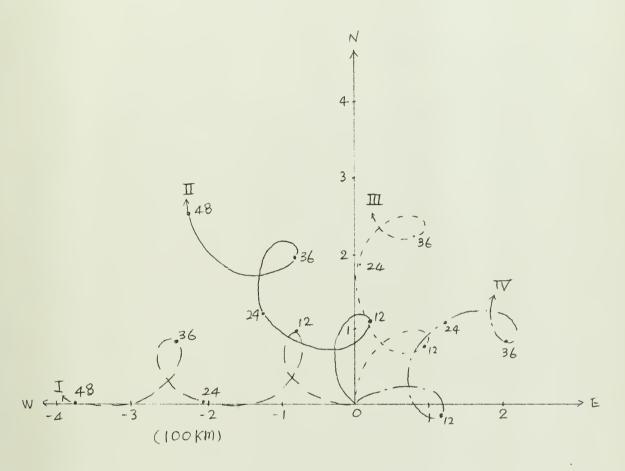


Figure 7

Figure 8 and Table II represent the trajectory for several initial velocities and basic pressure gradients.

Other conditions are similar to those in the previous model.

Curve I is for zero initial velocity and basic pressure gradient corresponding to a NW 10 kt geostrophic wind.

Curve II is for a 5 kt initial velocity and a 5 kt basic geostrophic wind.

Curve III is for a 10 kt initial velocity and



tasic sostrophic wind. The amplitude increases with increasing initial velocity and with increasing of the ma nitude of basic pressure gradient. The period does not change with changing of initial velocity and basic pressure radient.

TABLE II

1		I	I	III
Wo	hr-'	2	"	"
K	Km hr-'	1600	"	"
Фо	°lat. N	15	-	~
T,	km	300	9	
a	Kni	40	=	~
C _x	direction/kts (toward)	0	NW/5	NW/10
b _x	corresponding to Vgs.	NW/10	NW/5	NW/10

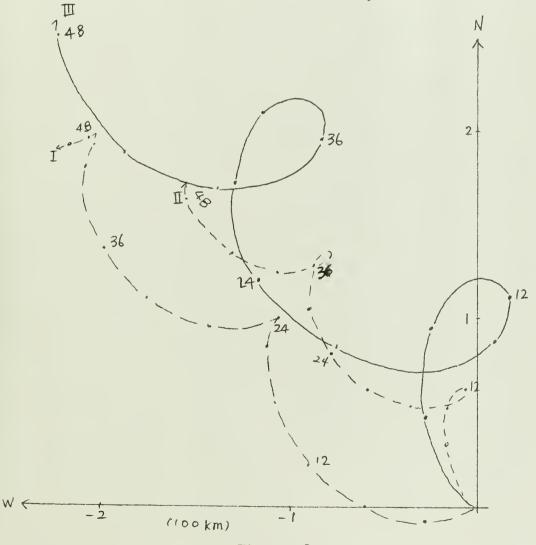


Figure 8



Figure 9 and TablelII are for an initial latitude of 25°N, but other data are the same as for the previous case. The results indicate that the amplitude and period decreases with increasing of initial latitude.

TABLE III

wo	K	r	8	Φ.	ъ	C
2 hr-1	1600 km2 hr-1	300 km	40 km	250N	NW/10 kt	s NW/10kts
	36	48		24	2 -	

Figure 10 and Table IV represent several cases of different vortex size and intensity. Curve III is same as II in the Figure 7. Curve I is for a case of large angular velocity of the rotating vortex. Curve II represents a large size and strong intensity (large angular velocity). The amplitude and period increase with decreasing intensity and increasing size.

-i Figure 9



TABLE IV

		I	II	III
wo	hr-1	4	"	2
K	Knithi-	6400	"	1600
Φο	°Lat N	15	"/	"
Y ₁	Km	300	600	300
a	KM	40	"	
C	direction Kts	NW/10	7	7
b (Vgs)	direction/Kts	NW/10	"	/-

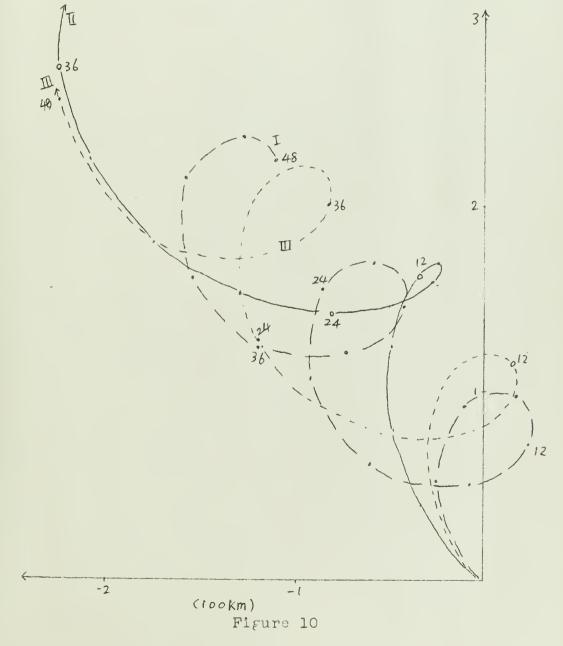




Figure 11 and Table V five the results for an elliptic pressure pattern which has a geostrophic wind velocity of 10 kts at x = 1110 kms y = 0, and 12.5 kts at x = 0, y = 1110 kms. The vortex model is taken as same as that of Figure 9. The initial velocity is taken as 80% of basic peosptrphic wind. The amplitude increase near the point of recurvature is due to increasin of pressure radient, and so the total speed decreases.

TABLE V

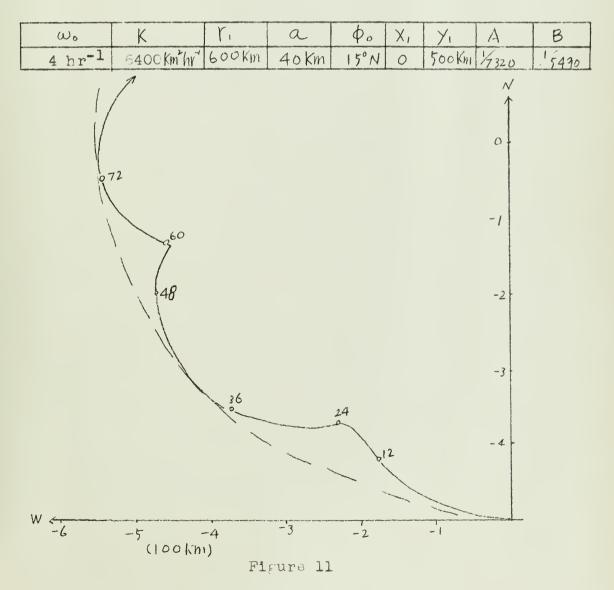


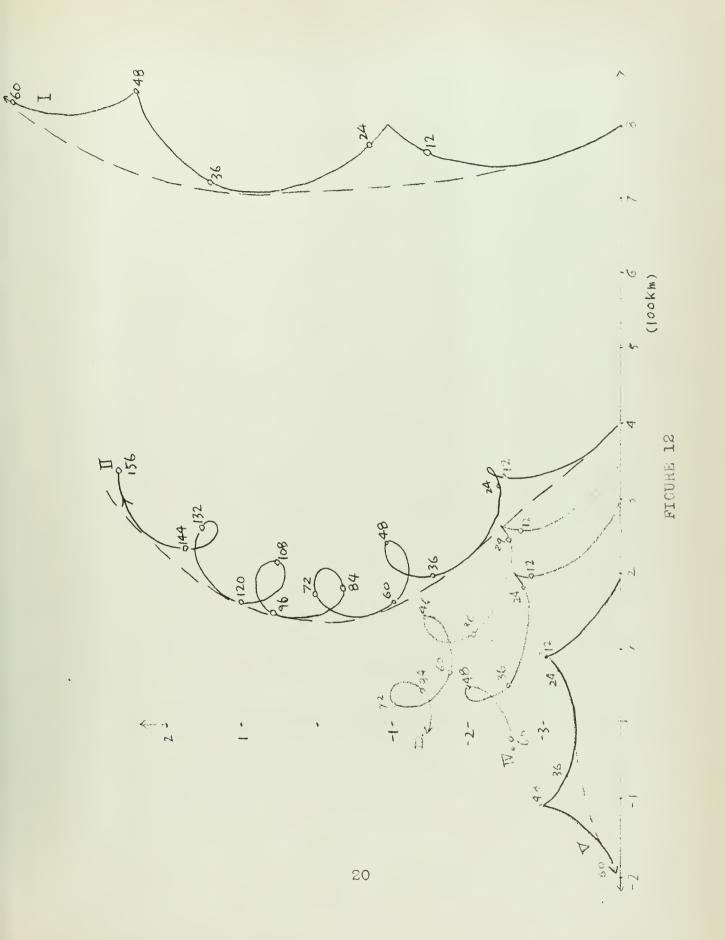


Figure 12 and Table VI are represented for the hyperbolic case. The vortex models are similar to those of Fig. 11, but each track is started from different position as shown in Fig. 12 and Table VI. Tracks I, II recurved toward NE after reaching the latitude where is $y = y_i$, Curves III, IV, V show continuous movement toward the west, together with a tendency southward according to the curvature of the basic isobars. The loop becomes lar e with crossin to the point of "col". Hence the overall velocity decreases accordingly. Curves II and III result from different starting points which differ by only 50 kms in x direction, but with same y-value. Yet one recurves and other does not. This 50 km distance is a small value compared with the size of the vortex; nevertheless it leads to significantly different results.

TABLE VI

	Y ₁ = 0	400Km hr 600 Km	-{	A = - B = -	7320	
		I	П	Ш	IV	∇
	X1 (Km)	- 800	-400	-350	-300	-200
I	yı (km)	+400	+400	+400	+400	+400







4. Verification

Time limitations did not permit adequate to tin of the theory, however two actual examples are shown for illustration. The first example is Typhoon "Clara" (Nov. 1950). The starting point is taken on Nov. 7, 1950, 0000 Japanese local time (1500Z 3th), in order to avoid the influence of a second vortex which occurred prior to this time. For prediction purposes, a promosis of the basic pressure field is needed; however, this type of promosis is not the subject of this paper. Therefore the time mean map which is the average of the 6th through the 10th will be used instead of a promostic map. Moreover, the 500-mb pattern will be taken as a basic pressure field. Figure 13 shows this pattern which may be represented by an elliptic paraboloid. The parameters A and D are as follows:

A = 1/14340, B = 1/9150the dashed track is for corrected A and B with A = 1/9150,

B = 1/6100 for the northern sector. The vortex data, obtained by averagin observed data is displayed in Table 7.

TABLE VII

C _{x0} =	5 kts	x, = 1100 kms
$C_{\gamma_0} =$	2.5 kts	y = 800 kms
r _i =	600 kms	φ _ε = 13.5° N
a =	40 kms	$\omega_{\rm o} = 4 \rm hr^{-1}$



Figure 13 Typhoon "Clara" Nov. 1950

22



Figure 14 shows the same example but with a variable A and B.

Assuming that A and B are linear functions of y only, A and B may be written as follows:

$$A \doteq A_{\circ} + (\frac{\partial A}{\partial y})y$$
 , $B \doteq B_{\circ} + (\frac{\partial B}{\partial y})y$

Here A, B, are initial values for A and B. For this example the values are taken as:

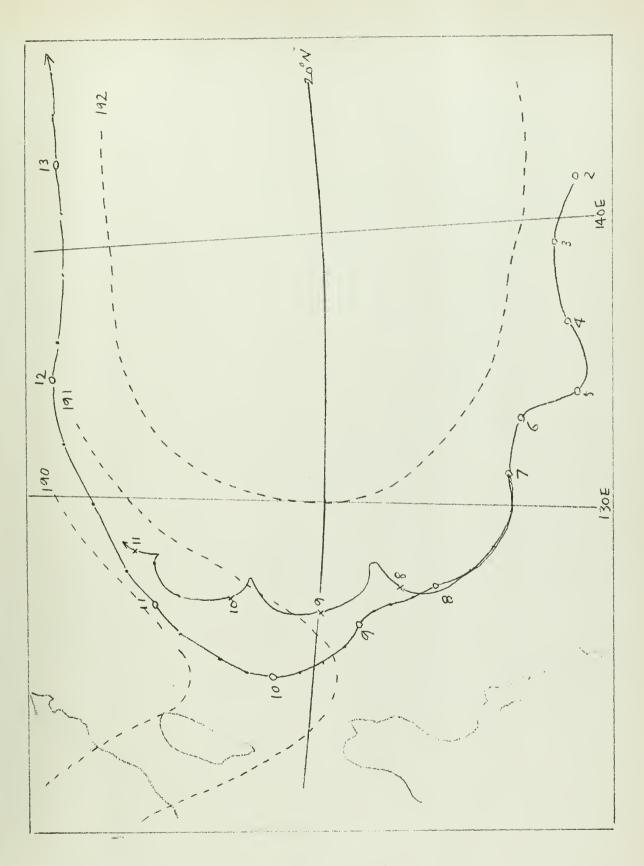
$$A_0 = \frac{1}{14640}$$
 $B_0 = \frac{1}{9150}$
 $\frac{\Delta A}{\Delta y} = \frac{1}{24400}/1600 \, \text{km}_s$
 $\frac{\Delta B}{\Delta y} = \frac{1}{12200}/1600 \, \text{km}_s$

Figure 15 also shows the same pattern, but represented by a hyperbolic paraboloid taking a similar variability for A and B as in the previous example. The values are taken as:

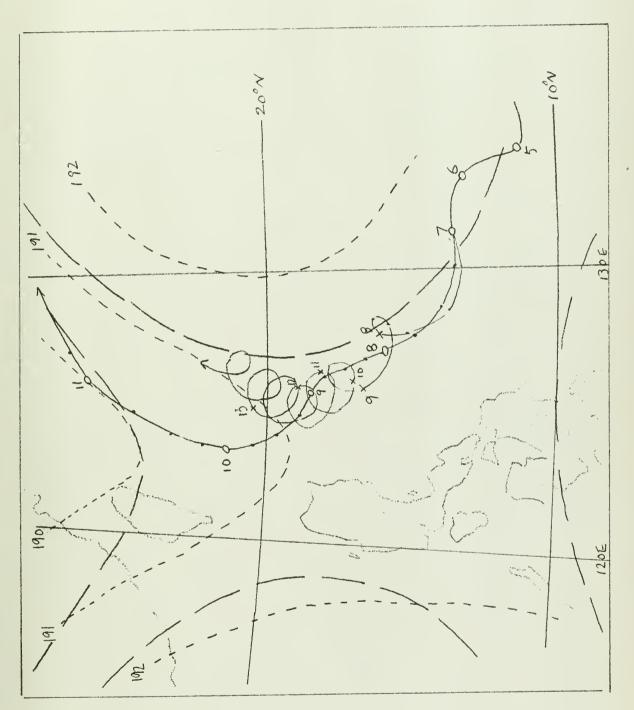
$$A_0 = \frac{1}{10460}$$
 $B_0 = \frac{1}{6655}$
 $\frac{\Delta A}{\Delta y} = \frac{1}{18300}/\frac{1600 \text{ km}}{1600 \text{ km}}$
 $\frac{\Delta B}{\Delta y} = \frac{1}{10460}/\frac{1600 \text{ km}}{1600 \text{ km}}$
 $x_1 = -800 \text{ km}$
 $y_1 = 550 \text{ km}$

The wortex, and other rata are the same as in Table VII.











The second example is for the Typroon "Jean" (Oct. 1956). The 5-day time-mean 500-mb map, representing the basic pressure pattern, is shown in Figure 16. The starting point is taken on 17th Oct. 1956, 0000 Japanese Local Time, (1500Z 16th). The mean 500-mb map seems to be the hyperbolic type. The parameters A and B are taken to be A = 1/4440, B = 1/4070, for track I, and A = 1/4070, B = 1/3660, for track II. The vortex model is shown in Table VIII.

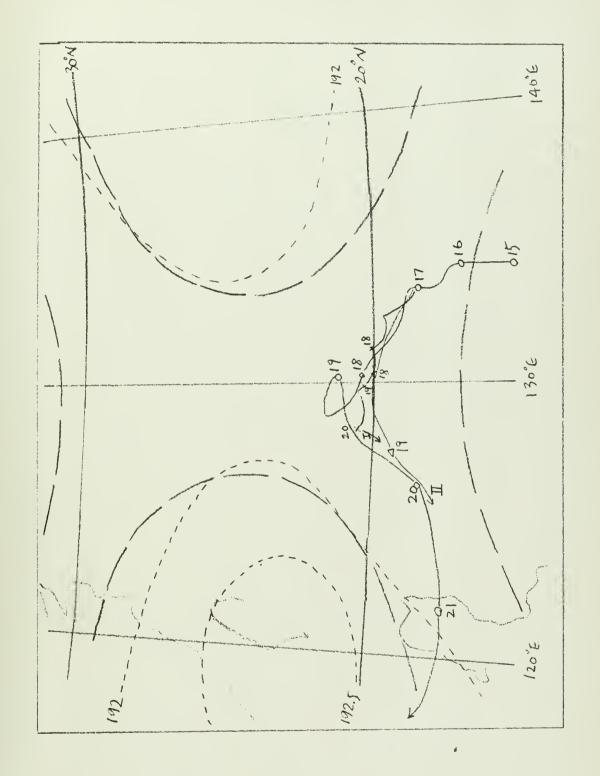
TABLE VIII

$\mathtt{C}_{\times_{\mathfrak{o}}}$	==	10 kts	\mathbf{x}_{I}	==	400 kms
Cyo	==	10 kts	У,		600 kms
r,	n miner	300 kms	Po	===	18.3 N
a	Tables or American	40 kms	ω_{o}	-200	2 hr-1

The computed track follows the actual trajectory in direction, quite well; however the computed speeds are sometimes slow or fast. This results from an under-or overestimate of the stren th of the basic pressure gradient.

Lack of time on the computer prevented further computations.







5. Conclusions

and may be considered to consist of an oscillatory motion superimposed on a steered motion, with the latter senerally dominating. Usually the amplitude of oscillation is small compared with the size of storms, hence it can not be recognized easily on daily maps. The oscillatory motion is superimposed on the steered rotion and the result is a curve similar to a trochoid or cycloid. The period of the oscillation has seen found theoretically to be 12/sin by many authors. In this study, the period was of the order of 1-2 days. The two examples of actual typhoons indicate that the results are applicable for a 3-5 day prediction of the trajectory of tropical storm, provided that the mean basic pressure field can be satisfactorily predicted.



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